

CNES Position Regarding the Use of the X- and Ka- Bands for EESS

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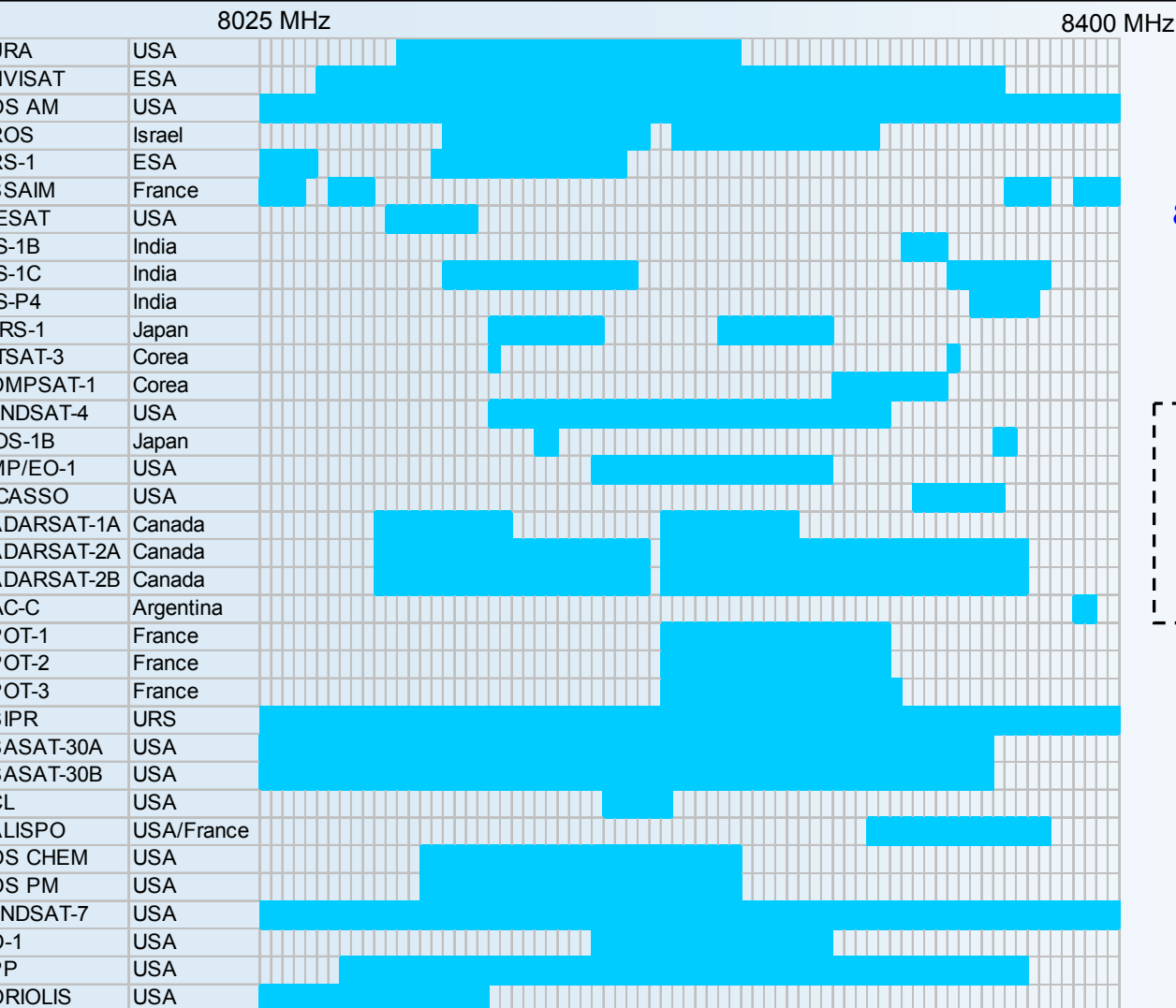
Data Rates Requirements

**Future earth observation space missions are based
on high resolution or large field optical and radar sensors
=> larger and larger amount of data are produced on-board satellites**

**Use of data compression mandatory,
but users want quasi-lossless compression
=> low compression ratio (between 3 to 6) only
=> high quality BER (Bit Error Rate) and FER (Frame Error Rate) required**
 $< 10^{-10}$ $< 10^{-7}$

**down-link data rates are increasing
but are limited by cost, mass and consumption**

Frequency Allocation



S-Band

available for low data rates

(< 2Mbps)

X-Band (8025 - 8400 MHz)

used by most of the data
telemetry payloads

Ka-Band (25.5 - 27 GHz)

allocated during WRC'97

Interference in X-Band

Two kinds of interference can occur :

- **In-band emissions interfere with other spacecraft using a part of the 8025-8400MHz band => degradation of both service**
 - geographic diversity of ground stations may help
 - spectrally-efficient coding and modulation may help
 - orbital coordination of EESS may help (like for GEO : frequency and position are jointly assigned)
 - Use of pointing on-board antenna
 - Migration to Ka-Band has to be considered
- **Out-of-band emissions interfere with other services using adjacent frequency bands => degradation of the other service (DSN for example)**
 - Filtered transmission can highly reduce the interference on other services
 - Use of pointing on-board antenna

In-Band Interference

CNES has explored some ways to reduce interference in X-Band :

- **Direct downlink in Ka-Band (25.5 - 27GHz)**
- **Inter-satellite link (to a relay hosted by a GEO)**
- **Use of large antenna for the ground stations**
- **Use of pointing on-board antenna**
- **Spectrally-efficient coding & modulation schemes**

Direct downlink at Ka-Band

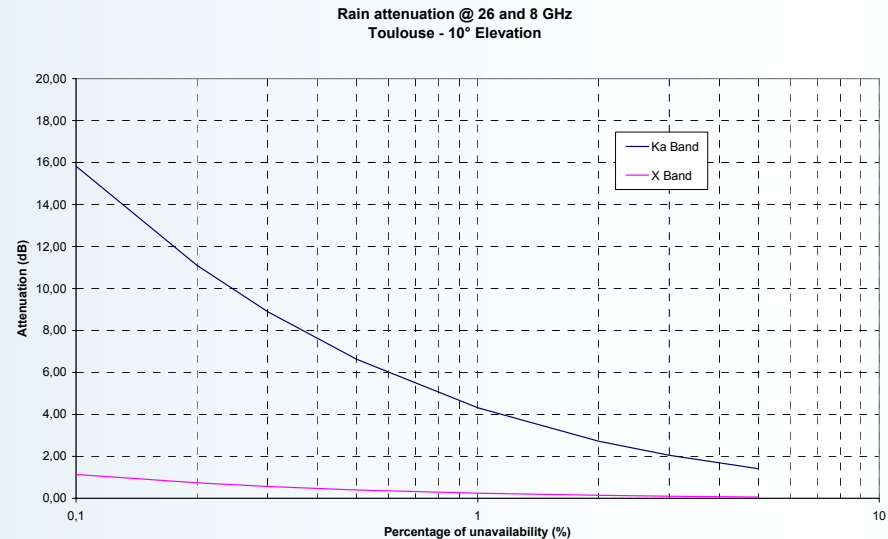
Use of the 25.5-27GHz Band

- **Propagation impairments**

- free space losses
- atmospheric gases, rain attenuation

- **Hardware**

- ground station antenna : gain increases
- on-board antenna
 - quasi iso flux \Rightarrow very low data rates, no ground station in rainy region
 - **pointing** \Rightarrow high data rates, link budget ensured everywhere, some mission constraints
- On-board amplifier : prototype developed by Thales (30W and 60W TWTA)
- Increase of the phase noise
- For some mission, it is **impossible** to track the satellite only **after 10° elevation** instead of 5° because of the resultant decrease of the number of crossing



Use of Ka-Band, 60GHz Band or optical link between the Earth Exploration Satellite and a geostationary satellite.

- No constraint on data flow congestion
- Improvement of mission performance
- **High complexity of the space segment**
- **High cost of the system development**

CNES study in 2002 :

- Use of a Ka-Band link with a GEO for 10% of mission data

Application in Europe :

- Silex, an optical link between Spot IV and Artemis
- Ka-band link between Envisat and Artemis

High latitude : nearly 12 crossings per day

- Esrange Satellite Station (Kiruna, Sweden) : *"We have not faced any problems on the X-band reception that has been related to interference from other satellites"*
- Need to use large antenna \Rightarrow narrow main and second lobes

Low latitude : nearly 5 crossings per day

- 3.5 m ground station used in Toulouse
- No data loss due to X-Band interference in Toulouse since 1986
- 5.4 m direct receiving stations all around the world (about 20 customers)

Use of broadcasting

- Highly increase probability of interference
- Transmission should be limited to periods when transmitting data to Earth stations (SFCG Rec 14-3R4)

Use of spectrally-efficient encoding & modulation schemes

Bandwidth is a scarce resource : a better use than today's one is necessary

⇒ Improve spectral efficiency over 1 bit/s/Hz
(actual common value)

Use of 8PSK modulation ⇒ Spectral occupation reduction of 1.25

(with 2.5/3 code rate versus RS coded QPSK)

Modulation	Efficiency (bits/symb)	Filtered efficiency (bits/s/Hz)
QPSK, RS coded	1.87	1.19
8PSK, RS + trellis 2/3	1.87	1.19
8PSK, RS + trellis 2.5/3	2.34	1.49

Note : this can be also considered as a 2.7 dB link budget improvement vs a RS coded QPSK modulation @ BER = 10^{-9}

8PSK On-Board Transmitter



Compact Size :
250 (with internal connection)
x 105 x 90 mm

Mass : around **1.4 kg**

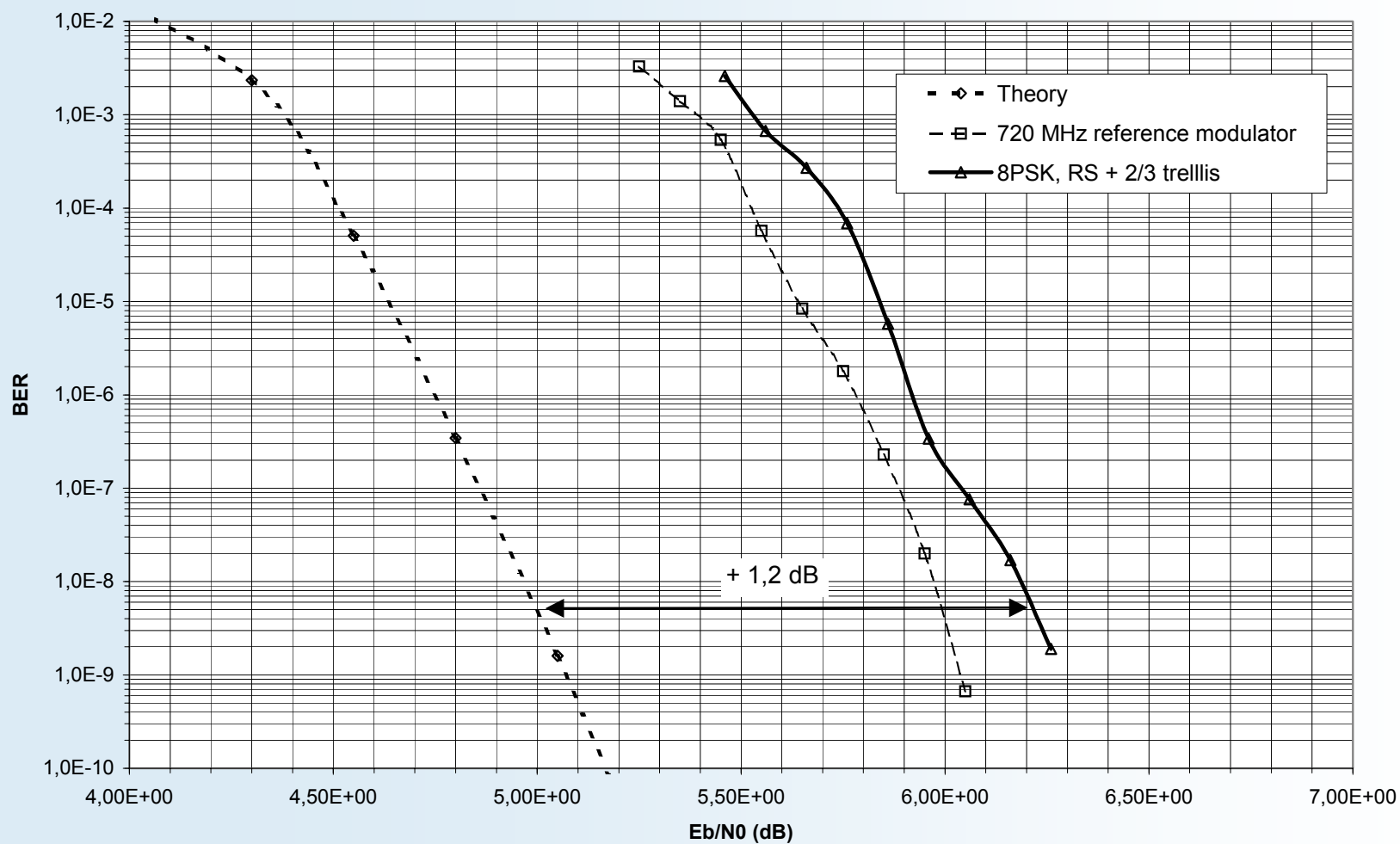
Consumption : around **30 W**

Effective Data Rate :
from **16.8 Mb/s**
up to **160 Mb/s**

Output Power : **8 dBW (EOL)**

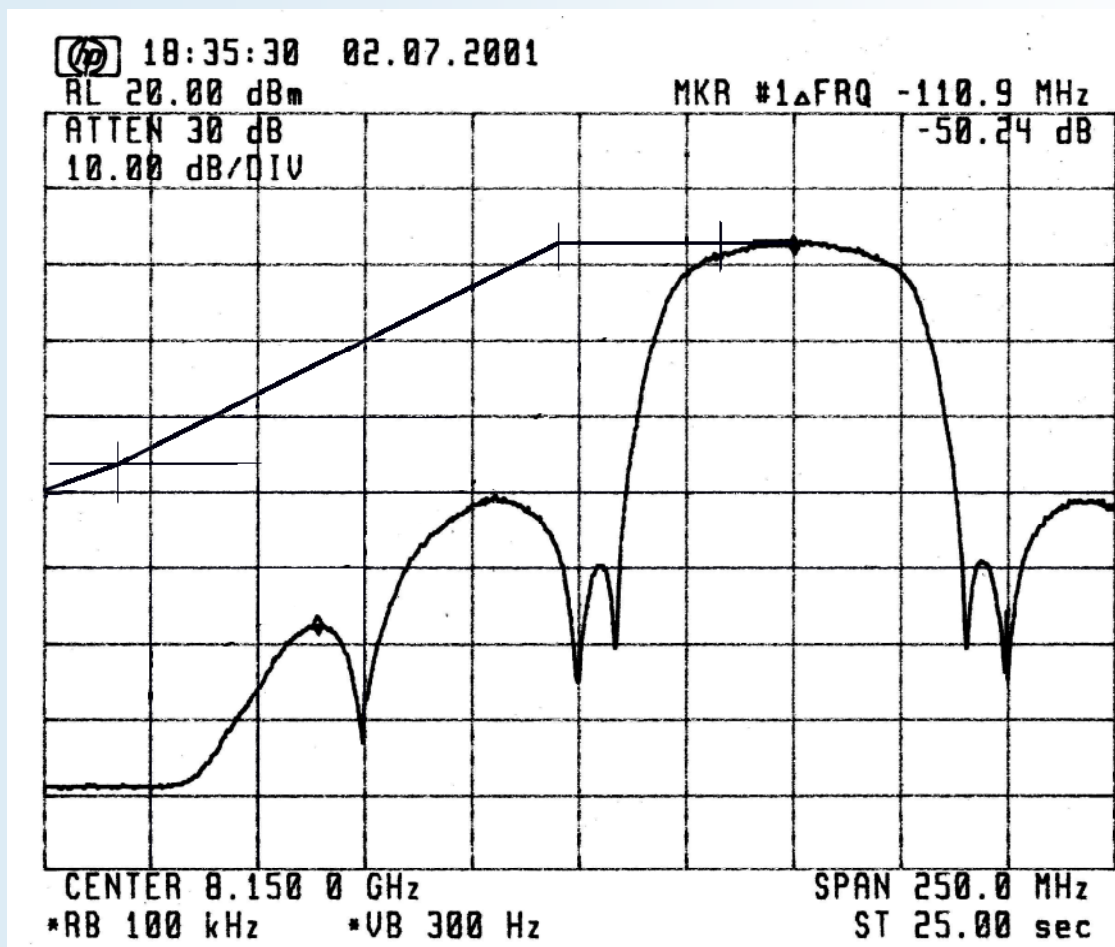
Failure Rate : **< 700 Fits @ 40°C**

Modem measured performance

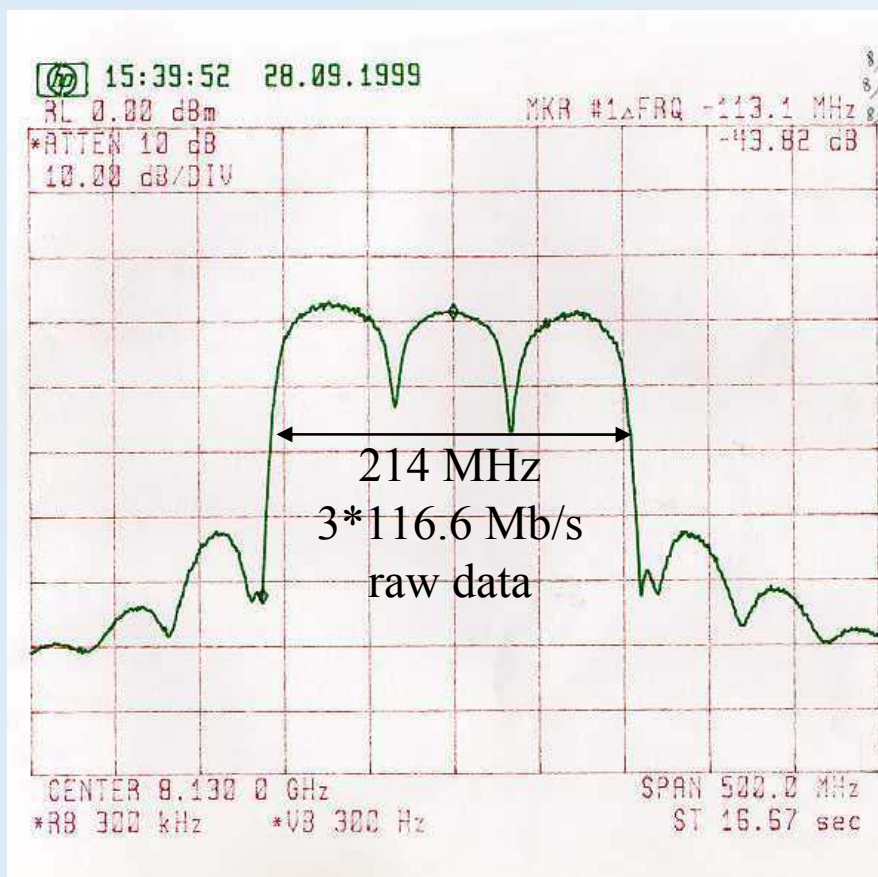


Output Spectrum at 50 MBauds

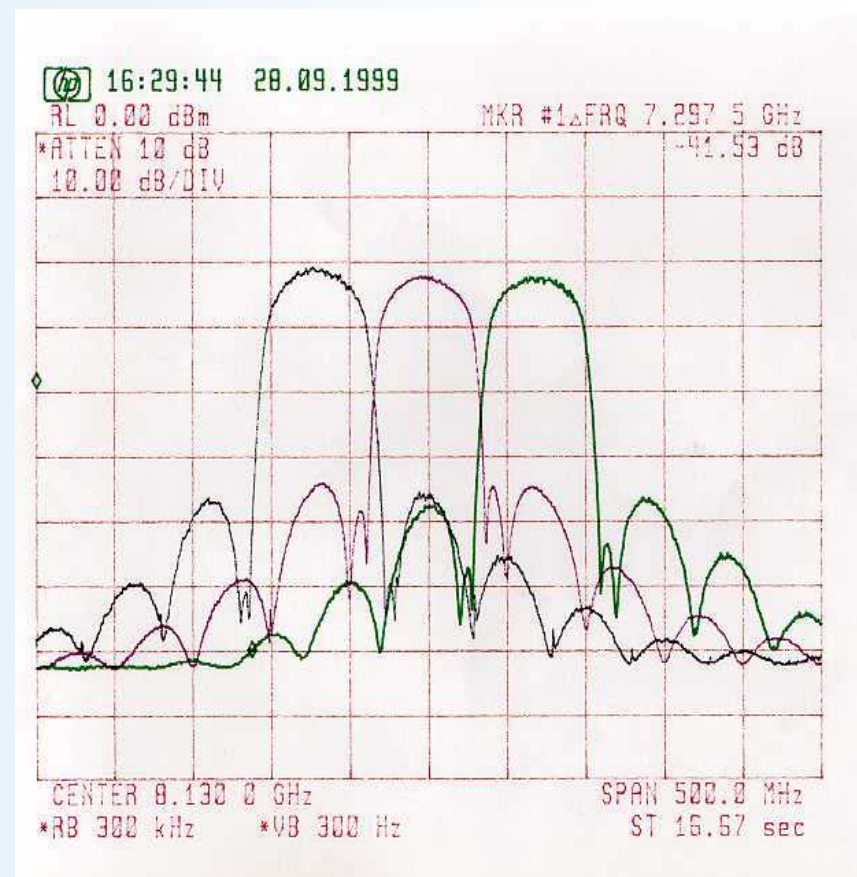
CCSDS 17-2R1



Spectra in quasi-SSRC multi-channel



3 channels at the OMUX output



The 3 channels combination

CCSDS 401.0-B "Radio Frequency and Modulation Systems—Part 1: Earth Stations and Spacecraft" (June 2001) recommends

that a mission planning to use conventional modulation methods which have an occupied bandwidth exceeding that permitted by the SFCG for the 8025-8400 MHz band, use 4D 8PSK TCM, provided that in no case shall the occupied bandwidth of said mission exceed that permitted by the SFCG

(2.4.18 : MODULATION METHODS AT HIGH SYMBOL RATES TRANSMISSIONS, EARTH EXPLORATION SATELLITES 8 GHZ BAND, SPACE-TO EARTH)

Use of spectrally-efficient encoding & modulation schemes

Applications of 8PSK modulation : on-going programs

- **CNES Demeter program**

- Detection of electro-magnetic emissions and earth radioelectric activity survey
- 8PSK/6.3W transmitter associated with a 3.5dBi isoflux antenna
- Protoflight models of the transmitter and the antenna delivered in 2002
- Satellite launch planned in 2004

- **CNES Pleiades program**

- Earth optical observation - Sub metric resolution
- Three 8PSK/30W transmitters, a steep edges OMUX and quasi iso-flux antenna
- Development of a 3 channels demonstrator to be tested end 2003
- Protoflight models to be delivered mid 2004

Use of spectrally-efficient encoding & modulation schemes

Applications of 8PSK modulation : future programs

- **Micro-satellite series for scientific and technologic programs**
 - Parasol (launch in 2004)
 - More than ten micro-satellites using 8PSK TCM telemetry will be launched during the next decade
- **Multi missions scientific programs such as ESA's SMOS program**

for additional information on 4D 8PSK TCM, contact Dr. Lesthievent
(Guy.Lesthievent@cnes.fr)

Today, no interference has been observed by CNES in the X-Band.

- Need to quantify with accuracy probability of interference between future Earth Exploration missions

Channelization of the 8025-8400 MHz

- certainly one of the worst way to use efficiently the band
- Need to consider separately the high and low latitude Earth station

Today, two simple solutions can be applied on the future missions

- Orbital/time coordination between agencies
- Do not use broadcasting mode

**If high data rate is needed, CNES recommends to use 4D 8PSK TCM
in order to reduce band occupancy and interference**

Computation of interference

- participation in the X-Band IWG (SFCG)
- Computation of interference and probability of interference with Matlab
- Accurate models in order to compute a worst case that can occur !

Spectrally-efficient modulation & coding

- High priority for CNES
- First flight planned in 2004

Direct downlink in Ka-Band

- No mission planned and no funds

Inter-Satellite Link

- No mission planned and no funds